

# Application of SpaceCube in a Space Flight System

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Note: This is the HANDOUT version of this presentation



# SpaceCube Development Team

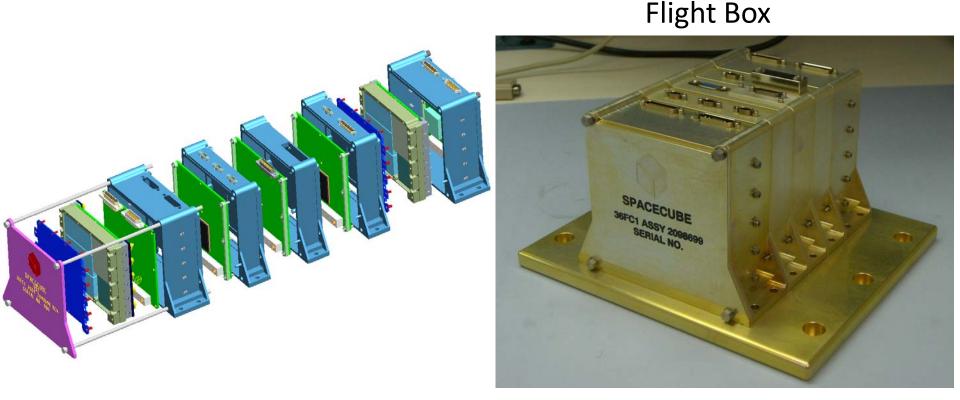
Mike Lin Walt Bradley **Dave Petrick** Gordon Seagrave John Godfrey **Beverly Settles** Tracy Price Gary Crum **Robin Ripley** Dan Espinosa Alessandro Geist **Dorian Seagrave** Wayne Greenwood Frank Cepollina Tom Flatley Giri Nadendla Madhu Kadari Bo Naasz Will Clement

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# GSFC SpaceCube

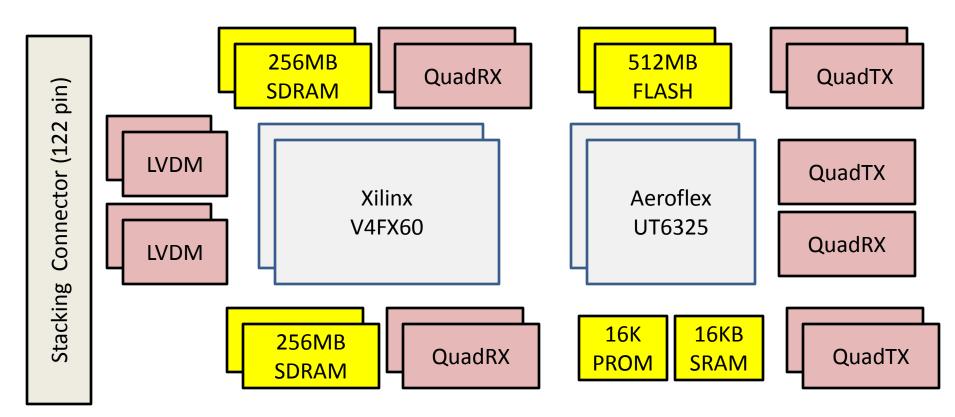
- Small, light-weight, reconfigurable multi-processor platform for space flight applications demanding extreme processing capabilities
- Based on Xilinx Virtex 4 FX60 FPGAs, 2 per processor card
- Stackable architecture



### Mechanical: 7.5-lbs, 5"x5"x7" Power: 37W (HST Application)

# SpaceCube Processor Card

- General: 4"x4" card, Back-to-Back FPGAs (x2), 7W typical power
- Memory: 1GB SDRAM, 1GB Flash, 16KB SRAM, 16KB PROM
  - Interfaces: 20 bi-dir differential signals, JTAG
  - Backplane: Power, 42 single-ended, 8 LVDM, 2 I2C, POR



# SpaceCube Processor Card

Top Side

**Bottom Side** 





# Hubble Servicing Mission 4 (STS-125)

- Relative Navigation Sensors (RNS) HST Payload
  - Record images of HST during docking and release, in particular the Soft Capture Mechanism
  - Perform on-orbit position and attitude estimation (Pose)
- RNS SpaceCube: Main Avionics Box
  - Controlled 3 cameras, GPS, 960GB memory, telemetry module, shuttle Ku downlink
  - Hosted Linux, VxWorks, C&DH, Automatic Gain and Integration Control, 2 pose image processing algorithms, TMR'd self-configuration scrubber using ICAP
  - Recorded GPS/AGC/POSE flight logs to flash
  - Consisted of 2 processors, 2 low-voltage power cards, 2 digital control cards, 1 JPEG2000 compression card

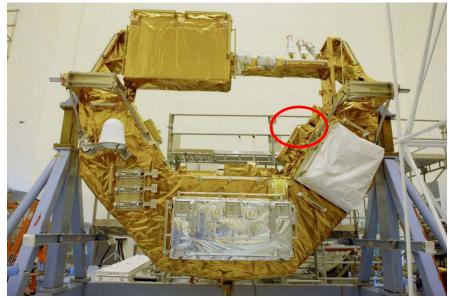
# **Relative Navigation Sensors**

• RNS originated from HST robotic service mission



 RNS operations conducted from JSC Space Shuttle Mission Control Center, Houston TX

## **RNS Hardware**



**MULE Carrier** 





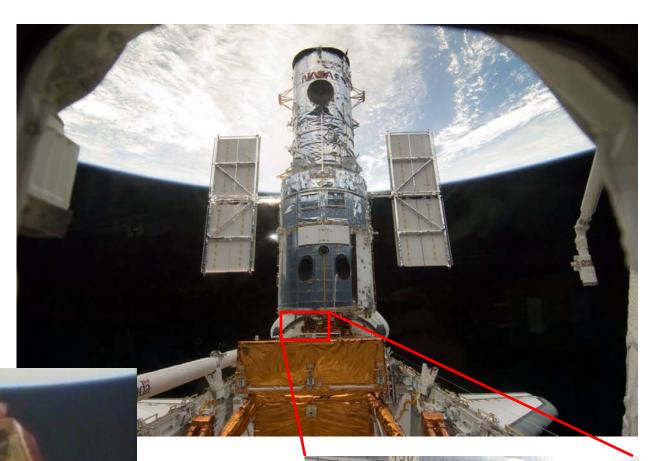
**RNS** Avionics Panel

MAPLD 2009 - Session A

STS-125 Payload Bay

# On Orbit

RNS Cameras

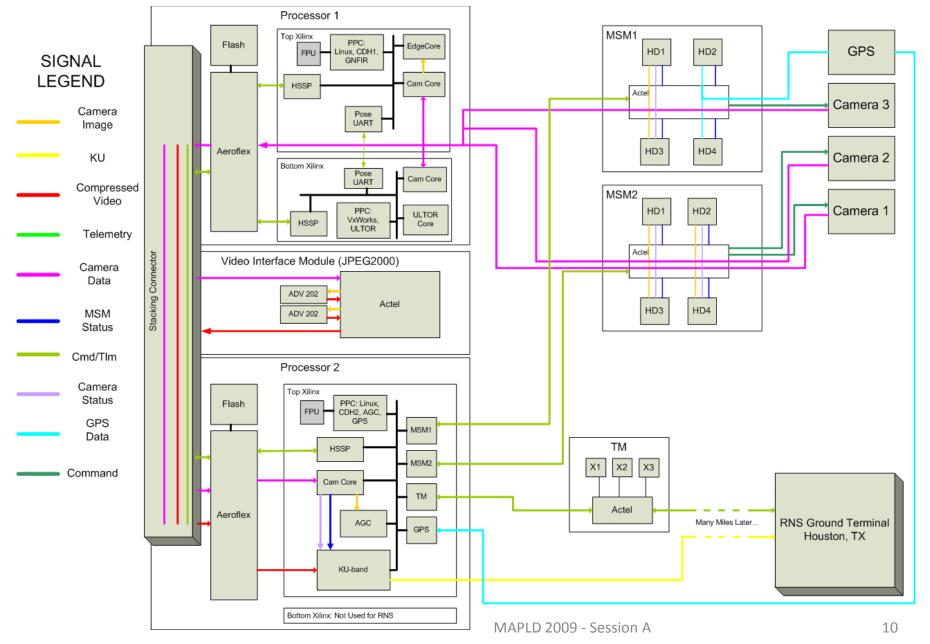


-C- ALC AV PAN -030.6 FOC 019.0 +15C CTV GAM BLK TLT -006.8 ZOOM 74.37 T/9 GMT 133:01:02:18 AUTO ALC SHOF SUN 0db Tough to see, but our cameras and SpaceCube are watching



MAPLD 2009 - Session A

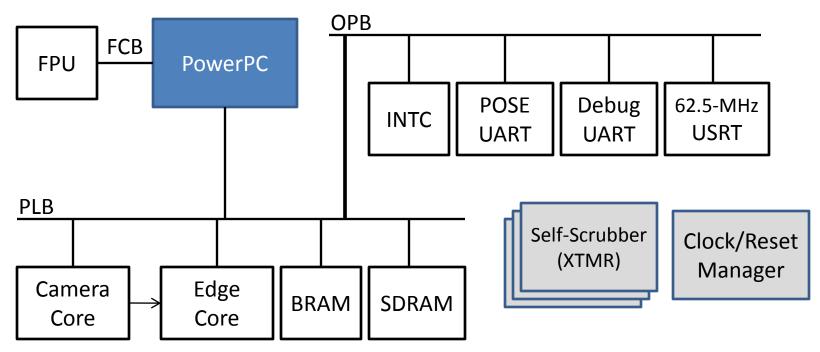
### SpaceCube Functional Block Diagram



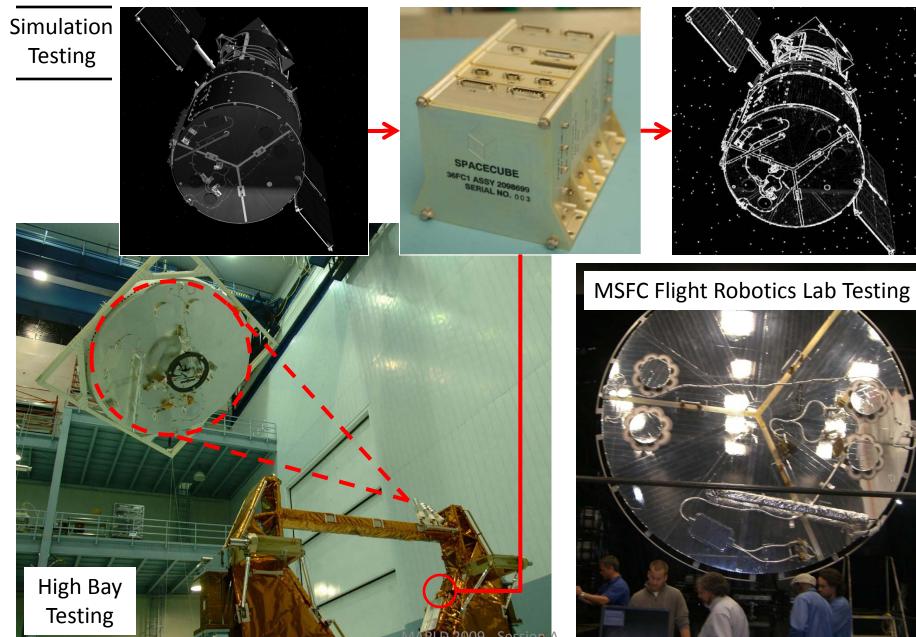
# Xilinx FPGA Design(s)

- Heavy PowerPC usage
- Logic: 62.5-125MHz, Processor: 250MHz
- Used 3 of 4 Xilinxs at 60-80% resource utilization

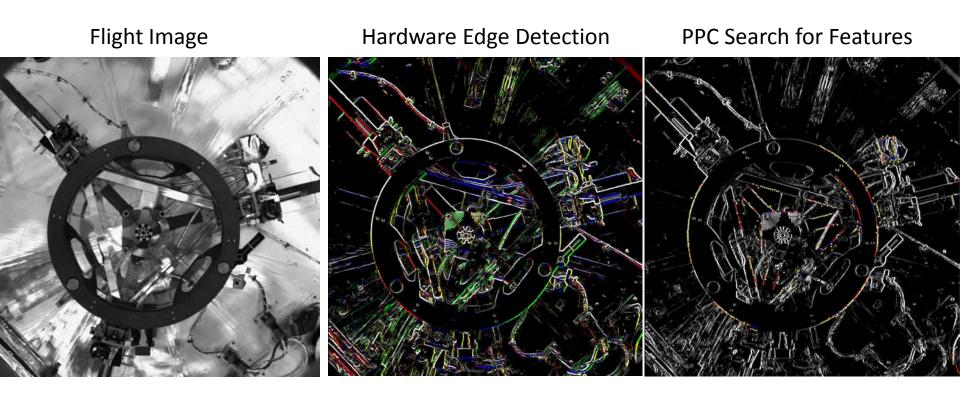
High-Level Example Xilinx Design (POSE #1 FPGA on Processor Card 1)



### **Barrage of System Testing**



# Tracking Algorithms on SpaceCube



# Tracking Algorithms on SpaceCube

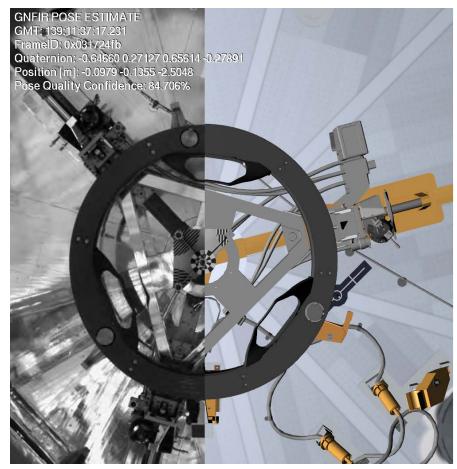
#### Long Range Camera on Rendezvous



**RNS Tracking Solution** 

Flight Image

#### Short Range Camera on Deploy



**RNS Tracking Solution** 

# **RNS Results Summary**

- SpaceCube enabled RNS to meet all objectives
  - Recorded 6 hours of camera and GPS data
  - Successfully tracked HST during rendezvous for 21 minutes and deploy for 15 minutes
  - Sent 100,000+ compressed images to ground
- SpaceCube powered for 60 hours (8 in SAA)
  - 2 configuration SEUs in SAA scrubbed out
  - 1 PowerPC SEE that watchdog repaired

# **HST Tracking**

Note: This is the HANDOUT version of this presentation, actual version contains movies



Deploy

Rendezvous

# What's Next?

- Just completed raw imagery downloads at GSFC
- Looking for another flight for RNS hardware
  Possible ISS flight for robotic demonstration
  - Hubble Robotic De-orbit Mission?
- Flight spare SpaceCube going to ISS (STS-129)
  Serve as NASA test bed for radiation mitigation
- Building two new versions of SpaceCube
  Covered in Session E presentation

## Questions?

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# Acronyms

- FPGA: Field Programmable Gate Array
- I2C: Inter-Integrated Circuit
- ICAP: Internal Configuration Access Port
- ISS: International Space Station
- LVDM: Low Voltage Differential Multi-drop
- MULE: Multi-Use Logistics Equipment
- POR: Power On Reset
- PPC: PowerPC
- SEE: Single Event Effect
- SEU: Single Event Upset
- TMR: Triple Module Redundancy